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1	Analytical and numerical studies on impact force profile of
2	RC beam under drop weight impact
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7	
8	Abstract: Impact force and structural response of reinforced concrete (RC) beams under drop
9	weight impact have been intensively studied and reported in the literature. The prediction of the
10	peak impact force has been well investigated while the study of the overall impact force profile
11	governing the structural response is limited. The impact force parameters acting on a structure
12	including the peak impact force, plateau force, and impact duration all affect the structural
13	responses. This study investigates the impact force profiles from drop weight impact on RC
14	beams. The effects of global stiffness (by changing the beam span and boundary condition) on
15	the impact force profile are numerically studied by using LS-DYNA. A two-degree-of-freedom
16	(2DOF) analytical model is developed for RC beams under drop weight impact by using fiber
17	beam section analysis method to predict the impact force. The developed 2DOF analytical
18	model is validated by drop weight impact test results. With the validated analytical model, a
19	simplified model of impact force profile is then proposed by determining six characteristic
20	points, which are expressed by empirical equations fitted by the analytical parametric study

- 21 results. The simplified impact force profile predicted by the empirical equations shows a good
- agreement with the test data.
- 23 Keywords: Analytical study; Numerical study; Impact force profile; Drop weight; Reinforced
- 24 concrete beams

25 Nomenclature

- A Contact area
- $A_{c,i}$ Area of concrete fiber
- $A_{s,i}$ Area of steel fiber
- *b* Width of beam section
- $c_{\rm b}$ Beam damping
- *c*_c Contact damping
- *d* Depth of beam section
- $F_{\rm a}$ Impact peak force
- $F_{\rm b}$ Resistance of RC beam under concentrated load
- F_{p} Impact plateau force
- $k_{\rm b}$ Global stiffness of beam
- $k_{\rm c}$ Contact stiffness
- *l* Beam span
- $l_{\rm eff}$ Effective beam span
- $l_{\rm p}$ Length of plastic hinge
- $m_{\rm b}$ Mass of RC beam
- $m_{\rm d}$ Mass of drop weight
- *M* Bending moment of beam section
- N Axial force of beam section
- $P_{\rm dyn}^{\rm max}$ Dynamic shear capacity of beam section
- t_0 Duration of local response
- $t_{\rm a}$ Time at peak force
- $t_{\rm b}$ Duration of primary impact
- t_{cd} Duration of force plateau
- $t_{\rm e}$ Total impact duration
- u_i Displacement in governing equations
- \dot{u}_i Velocity in governing equations
- \ddot{u}_i Acceleration in governing equations
- v_0 Initial velocity of drop weight
- $v_{\rm b}$ Velocity of beam at midspan
- $v_{\rm d}$ Velocity of drop weight
- $Z_{c,i}$ Coordinate of concrete fiber along z-axis
- $Z_{s,i}$ Coordinate of steel fiber along z-axis
- β Scale factor for contact stiffness
- δ Deflection of beam section
- $\dot{\delta}$ Deflection rate of beam section
- $\delta_{\rm b}$ Deflection of RC beam at midspan
- $\varepsilon_{c,i}$ Strain of concrete fiber
- $\dot{\varepsilon}_{c,i}$ Strain rate of concrete fiber
- ε_N Section strain at the centroidal y-axis
- $\varepsilon_{s,j}$ Strain of steel fiber
- $\dot{\varepsilon}_{s,j}$ Strain rate of steel fiber
- $\sigma_{c,i}$ Stress of concrete fiber
- $\sigma_{s,j}$ Stress of steel fiber
- ϕ Curvature of beam section

$\dot{\phi}$ Curvature rate of beam section

26 **1. Introduction**

27 The study on reinforcement concrete (RC) beam under impact has drawn interests of many researchers due to the increasing numbers of natural or man-induced extreme events [1-3]. Drop 28 29 weight impact test is a widely used approach to examine the impact behaviour of RC beams. 30 Upon drop weight impact, the impact force between a drop weight and an impacted beam is 31 measured by a load cell and the displacement at midspan is recorded to examine the dynamic 32 response of beams. RC beams under impact load may experience flexure failure, flexure-shear 33 failure, or local punching shear failure, depending on the impact loading rates, impact load-34 carrying capacities of beam section and peak impact force [4-6]. The peak impact force is 35 primarily governed by the local contact stiffness of impact zone and impact energy [7]. 36 Moreover, the drop weight impact results in stress waves propagating from the impact point to 37 the beam ends during the local response phase, leading to a change of the effective beam span 38 [8, 9]. Once the whole beam span is mobilized, the beam enters the global response phase and 39 the global stiffness would affect the impact force [10]. Therefore, the impact force influenced 40 by different factors determines the dynamic response and failure of RC beams subjected to drop 41 weight impact.

42 Impact force profiles of RC beams subjected to drop weight impacts can be categorized 43 into three types [10]. It is found that the classification of the impact force profile type depends 44 on the ratio of drop weight mass to beam mass. Various mass ratios result in different relative 45 velocities between a drop weight and an impacted beam, causing different interactions at the

46	impact zone. To damage RC beams, heavy impact mass is used in the drop weight test and thus
47	the impact force profiles with force plateau (i.e., Type III impact force profile) are often
48	observed in impact test results. As summarized in Fig. 1, the impact force values measured in
49	the drop weight impact test are affected by different factors. Increasing impact energy causes a
50	higher primary peak force [11, 12]. It is worth noting that impact velocity has a more significant
51	effect on the primary peak force than impact mass [7]. Moreover, the contact stiffness, relating
52	to the drop weight test setup (e.g. impact interlayer and drop weight geometry) and contact
53	material property, also affects the peak impact force [13]. The existence of an initial inclination
54	of drop weight and the drop weight head with a smaller radius decrease the contact areas and
55	thus reduce the peak impact force. The stiffer interlayer such as steel plate or load cell placed
56	between a drop weight and a beam increases the contact stiffness, which causes a higher impact
57	force. Although there are many studies on the peak impact force through experiments and
58	numerical simulations [12, 14-16], very limited systematic studies investigating the influencing
59	factors on the impact force plateau have been conducted. A previous study [10] found that the
60	force plateau is related to the global stiffness of beam after the primary impact pulse. Therefore,
61	the factors influencing the global stiffness such as the beam section, beam span, rebar ratio, and
62	boundary condition should be investigated to examine their effects on the impact force plateau.
63	It should be noted that the structural response such as the maximum displacement usually
64	appears at the end of impact force plateau. Therefore, an accurate prediction of impact force
65	profile is essential for structural response analysis.



Fig. 1. Factors affecting the impact force profile

66 67

68 Different methods have been employed to study the impact behaviour of RC beam under 69 drop weight impact, i.e., experimental, numerical, and analytical methods [17-22]. The 70 experimental tests allow direct measurement and observation of the impact behaviour of beams. 71 However, it is hard to compare the results from different studies due to different test setups and 72 due to the highly nonlinear nature of results, which causes the results may not be extrapolated 73 [4, 8, 13]. The numerical method provides a technique to predict the impact behaviours of 74 various beams with different test setups. Stress wave propagation along the beam and more 75 results such as stress, strain, and acceleration can be easily retrieved and compared from 76 numerical results [7, 23, 24]. However, it is time-consuming to develop and run high-fidelity 77 numerical models. The analytical methods based on the mass-spring system (SDOF and 2DOF) 78 have been used for the predictions of dynamic responses of beams under impact loads [18, 21, 79 25]. The impact force and the global response can be obtained by theoretical derivations with 80 some idealizations and assumptions. It should be noted that reasonable assumptions about the

81 characteristics of impact behaviour are very important for the accuracy of the analytical 82 predictions. However, the stress wave propagation along beam, effective beam span, and 83 detailed calculation of global stiffness are not straightforward, hence were not well considered 84 in previous analytical studies.

85 To date, impact force profile of RC beams under impact loads has not been well studied 86 while the analytical method for fast and accurate prediction on the impact force profile of RC 87 beam is not yet available. In this study, the numerical model is developed in LS-DYNA and 88 calibrated with the testing data of RC beams under drop weight impact. The verified numerical 89 models are employed to study and to quantify the effect of the global stiffness of RC beams, 90 i.e., beam span and boundary condition, on the impact force plateau. Moreover, a 2DOF 91 analytical model is developed for RC beams under drop weight impact by considering stress 92 wave propagation. The developed 2DOF analytical model is validated by the drop weight 93 impact test results. With the validated analytical model, large amounts of impact force data are 94 generated for the development of a simplified impact force profile model. Empirical equations 95 for the simplified impact force profile model are proposed by considering the key factors 96 affecting the impact force. The impact force profiles predicted by the proposed equations are 97 also compared with the test results from other studies to further verify its accuracy.

98 2. Previous drop weight impact tests

99 To calibrate the numerical model of RC beam under impact loads, three specimens as listed
100 in Table. 1 were selected from the drop weight impact tests conducted by Fujikake et al. [26].

The drop weight impact setup and the dimension of the RC beam are shown in Fig. 2. The pinsupported RC beam was impacted by a steel drop weight with a mass of 400 kg falling from a height of 1.2 m. A total of four longitudinal rebars are placed in the beam section and the arrangement of longitudinal rebar for three beams is present in Table. 1. The 10 mm-diameter stirrups were placed along the beam at a space of 75 mm. The detailed material strength of concrete and steel rebar can be found in Ref. [26] for each specimen.



Table 1. Arrangement of longitudinal rebar [2	6].
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Sussimon	Compression side		Tension side		
specimen	Number and diameter (mm)	Area A_{s} (mm ²)	Number and diameter (mm)	Area $A_{\rm s}$ (mm ²)	
S1616-1.2	2*D16	397	2*D16	397	
S2222-1.2	2*D22	774	2*D22	774	
S1322-1.2	2*D13	126.7	2*D22	774	

108



109 110

Fig. 2. Test setup and dimension of RC beam (unit: mm) [26].

111 **3.** Numerical study of the impact force profile

112 **3.1. Numerical model and calibration**

113 **3.1.1. Numerical model**

114 The numerical model is developed in LS-DYNA. The concrete beam and drop weight are

simulated by 8-node solid elements. The non-linear beam elements are used to simulate the longitudinal rebars and stirrups. Penalty-based surface-to-surface contact is assigned between the concrete beam and drop weight. The interaction between concrete and reinforcements is defined by the constraint-based coupling method. The mesh size of 10 mm for the numerical model is adopted by conducting a mesh convergence study to obtain reliable results with reasonable computational time.

The concrete material is modelled by the concrete damage model (Mat_72R3) and the piecewise linear plasticity steel model (MAT_24) is employed to simulate the steel longitudinal rebars and stirrups, as commonly used in the numerical simulations [27, 28]. To consider the strain rate effect of concrete and steel under impact loading, the dynamic increase factors (DIFs) are defined for concrete material and steel material, respectively [29, 30]. In addition, the steel drop weight is modelled by the elastic material model (MAT_1).

127 **3.1.2.** Comparisons between numerical and test results

Three specimens S1616-1.2, S2222-1.2, S1322-1.2 with different longitudinal rebar ratios in Ref. [26] are employed to calibrate the numerical model. The drop weight falls from a height of 1.2 m, generating a velocity of 4.85 m/s. The numerical result is compared with the test results to validate the numerical models in terms of the impact force and displacement at midspan as shown in Fig. 3. It is found the predicted impact force profiles and displacements at midspan agree well with the test results. Therefore, the developed numerical model is proven yielding reliable predictions of the responses of RC beams subjected to drop weight impact.



135 **3.2.** Numerical study of the global stiffness effect on the impact force profile

136 It has been demonstrated in a previous study that global stiffness affects the force plateau of the impact force profile [10]. By assuming the global stiffness of a beam as a spring element 137 138 $k_{\rm b}$, the beam under drop weight impact can be simplified as a mass-spring system as shown in 139 Fig. 4. At the end of the primary impulse, the velocity of beam reaches its maximum and begins 140 to decrease due to the resistance provided by the global stiffness of the beam. The relative 141 velocity between drop weight and beam becomes smaller and the velocity and displacement of 142 beam at midspan and drop weight are comparable during the stage of impact force plateau, 143 which causes the midspan part of the beam and drop weight moving together [10]. When the global stiffness k_b in Fig. 4 is greater, the interaction between the drop weight and the beam 144 would be more intensive, which results in a higher impact force plateau. The global flexural 145

stiffness of beam is governed by the flexural stiffness *EI*, beam span *l*, and boundary condition,
which can be expressed in Eq. (1) and Eq. (2) for pinned and fixed end, respectively. The effects
of beam span and boundary condition on the impact force plateau are analyzed in the subsequent
sections.

For pinned end
$$k_{\rm b} = \frac{48EI}{l^3}$$
 (1)

For fixed end
$$k_{\rm b} = \frac{192EI}{l^3}$$
 (2)





153 **3.2.1. Effect of the beam span**

The effect of the beam span on the impact force profile is studied in this section. Five beams with the beam span of 1.4 m, 2.1 m, 2.8 m, 3.5 m, and 4.2 m are considered. The beams have the same section dimension and rebar layout as specimen S1616-1.2 as shown in Fig. 2. Two sets of impact scenarios are considered as given in Table 2, which includes the beams subjected to impact with a constant impact mass ratio of 3.0 (the beam mass increases with the

159 be	eam span)	and the beams	impacted b	y a constant im	pact mass of 400 kg.
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1	l'able 2. Impact ma	ass and mass ratio.		
Impact scenario	Beam span / m	Impact mass / kg	Beam mass / kg	Mass ratio
	1.4	400.0	131.3	
	2.1	590.6	196.9	
Impact with constant mass ratio	2.8	787.5	262.5	3.0
	3.5	984.4	328.1	
	4.2	1181.3	393.8	
	1.4		131.3	3.0
	2.1		196.9	2.0
Impact with constant mass	2.8	400	262.5	1.5
	3.5		328.1	1.2
	4.2		393.8	1.0

Table 2 Impact mass and mass ratio

161 3.2.1.1. Impacted by a constant mass ratio

Increasing the beam span inevitably causes the increase of beam mass, which results in a decrease of mass ratio of drop weight to beam given the same impact mass. Since the mass ratio affects the impact force profile, a constant mass ratio of 3.0 is adopted first in this study to investigate the effect of span on impact force plateau. The impact mass and mass ratio for the specimens with different clear spans are listed in Table 2.

167 The impact force of the beams under a constant impact mass ratio of 3.0 is illustrated in 168 Fig. 5. As shown, in the primary force stage, the peak impact force F_a increases from 302.06 169 kN to 405.22 kN due to the increase of impact mass (i.e. impact energy). After the primary 170 pulse, there are force fluctuations due to the interaction between the drop weight and the beam. 171 After the transitional stage, the drop weight and beam move together and the relative velocity 172 between them is almost zero. The impact force keeps an almost constant value F_p that can be 173 defined as the average impact force during the plateau stage. The plateau force F_p drops from 93.17 kN to 31.90 kN, with a decrease of 65.76%, as the beam span increases from 1.4 m to 4.2
m, but the corresponding duration of the plateau stage becomes longer with the increased beam
span.



177 178



¹⁷⁹ *3.2.1.2. Impacted by a constant mass*

To further investigate the effect of the beam span on the impact force profile, five beams are impacted by the drop weight with a constant mass of 400 kg (i.e. the mass ratio varied from 3.0 to 1.0 as listed in Table 2). The impact force time history is presented in Fig. 6. The primary force peaks of the specimens are similar, indicating that the span has negligible effect on the primary force peak, which has been also verified by Pham and Hao [31]. In the plateau stage, however, the plateau force decreases by 72.07%, from 93.17 kN to 26.02 kN. With the increase 186 of beam span, the duration of the plateau stage becomes longer. Therefore, increasing the beam 187 span causes decrease of plateau force and increase of plateau stage duration due to the decrease









191 **3.2.2. Effect of the boundary condition**

The responses of RC beams with pinned and fixed boundary conditions are compared as shown in Fig. 7. The effective span of 2.8 m, 3.5 m, and 4.2 m are considered. The impact velocity of the drop weight is 4.85 m/s. The mass ratio of drop weight to beam remains 3.0 and the impact mass is listed in Table 2 for the beams with different spans. The time histories of impact force are shown in Fig. 8. Even though the beams with the same span have different boundary conditions in each group, the primary force peaks are similar as shown in Fig. 8(a)~(c). 198 It indicates that the primary force peaks are independent of the boundary conditions. This result 199 is supported by the previous study [31] which showed the boundary condition does not affect 200 the peak impact force. The peak impact force depends on the impact mass, impact velocity, and 201 contact stiffness [13]. A higher plateau force appears in the fixed end beam than the pin-202 supported beam. For example, for the beams with a span of 2.8 m as shown in Fig 8(a), the 203 plateau force of the fixed beam (141.02 kN) is larger than that of the pin-supported beam (47.14







205



Fig. 8. Impact force acting on the beams with different boundary conditions.

206 **3.2.3.** Discussion of the global stiffness effect on the impact force profile

207 Given the same input impact energy, the primary force peaks are similar for the specimens 208 regardless of the beam clear span and boundary condition, which agree with the results in Refs. [4, 31]. However, the plateau force decreases with the reduction of beam global stiffness (i.e. 209 210 increasing the beam span or using the less-constrained boundary condition), but duration 211 increases. Moreover, by analyzing the test results presented in Fig. 3, the specimen S2222-1.2 212 with a higher tensile reinforcement ratio of 2.02% experiences a larger plateau impact force of 213 156.18 kN because of its larger global stiffness, compared to that of 95.66 kN in the specimens 214 S1616-1.2 with a tensile reinforcement ratio of 1.07%. Therefore, it is evident that the beam 215 with lower global stiffness provides lower resistance which weakens the interaction between 216 the drop weight and beam and thus causes a lower plateau force. The duration of the force 217 plateau becomes longer for the specimen with lower global stiffness as shown in Fig. 3, Fig. 5, 218 Fig. 6, and Fig. 8. This is because the lower global stiffness leads to a less beam resistance to 219 stop the downward movement of the beam and the drop weight. A longer time is needed to 220 reduce the velocity of the beam to zero and to reach its maximum midspan displacement. It is 221 concluded that the global stiffness has negligible effect on the primary force peak but has 222 significant influence on the plateau force. Based on this conclusion and other studies on the 223 factors affecting the impact force [10, 13], a simplified impact force profile model is proposed 224 by considering these influencing factors in the subsequent analytical study.

4. Analytical study of the impact force profile

226 **4.1. Analytical model**

227 **4.1.1. 2DOF model**

228 To develop a proper simplified impact force profile model, various factors affecting impact 229 force as listed in Fig. 1 are considered in the analytical model based on a 2DOF mass-spring-230 damping system. Since the impact response consists of the local response at the impact zone 231 and the global response of a beam [12, 18, 31], the pin-supported beam under drop weight 232 impact can be simplified as the 2DOF mass-spring-damping system as shown in Fig.9. The 233 drop weight and beam are assumed as two lumped masses, and the local contact stiffness and 234 the global stiffness of beam are assumed as two springs. The analytical model is explained in 235 detail in the following section and the applicability of the model is discussed in section 4.4.





Fig. 9. 2DOF model of an RC beam under drop weight impact.

238 The governing equations of the 2DOF mass-spring-damping system are expressed as
239 follows:

$$m_d \ddot{u}_d + c_c (\dot{u}_d - \dot{u}_b) + k_c (u_d - u_b) = 0$$
(3)

$$m_b \ddot{u}_b + c_b \dot{u}_b + c_c (\dot{u}_b - \dot{u}_d) + k_b u_b + k_c (u_b - u_d) = 0$$
(4)

where u_i , \dot{u}_i , and \ddot{u}_i (*i* = *d* or *b*) are the displacement, velocity, and acceleration of the drop 240 241 weight or beam mass, respectively. The parameters include the initial impact velocity of the 242 drop weight v_0 , drop weight mass m_d , contact stiffness k_c , contact damping c_c , effective mass of 243 the beam m_b , global stiffness of the beam k_b , and beam damping c_b . Herein, the initial impact 244 velocity v_0 and drop weight mass m_d can be determined by the designed impact energy. 245 Moreover, selecting an appropriate contact law is important to predict impact force for the local 246 contact of impact. Based on the Hertzian contact theory, pure elastic contact model and 247 dissipative visco-elastic contact model were developed [32-34]. The dissipative visco-elastic 248 contact model considers the energy loss during the impact by using the contact damping and it 249 is used to predict contact force for impact actions [35, 36]. In this study, the contact stiffness 250 and contact damping are incorporated into the analytical 2DOF model as shown in Fig. 9. The 251 contact stiffness (k_c) is related to the contact area, drop weight head geometry, and contact 252 material properties [13, 37, 38]. In regard to a flat head, the contact stiffness is calculated by 253 using an equation similar to the penalty contact algorithm in LS-DYNA [39] as follows:

$$k_c = \beta \frac{EA}{h} \tag{5}$$

254 where *E* and *A* are Young's modulus of concrete and contact area, respectively, *h* is the thickness 255 of the contact area, β is a scale factor to adjust the contact stiffness by calibrating the impact force. If the drop weight head has a curved shape, the contact stiffness is defined by using the
Hertz contact model [32, 40]. The contact stiffness is expressed as follows:

$$k_{c} = \beta \frac{4\sqrt{R_{h}}}{3} \left(\frac{1 - \nu_{d}^{2}}{E_{d}} + \frac{1 - \nu_{b}^{2}}{E_{b}} \right)^{-1}$$
(6)

where R_h is the radius of the curved head, E_d and E_b are Young's modulus of drop weight and beam, respectively, v_d and v_b are Poisson's ratio of drop weight and beam, respectively. Moreover, the contact damping should be considered to avoid the intensive oscillation of the impact force [20, 41]. The contact damping c_c proposed by Anagnostopoulos [42] is adopted and expressed as follows:

$$c_c = 2\xi_c \sqrt{\frac{m_d m_b k_c}{(m_d + m_b)}} \tag{7}$$

where ξ_c is the contact damping ratio and is determined as 0.2 by calibrating the test results, c_b is the damping of RC beam for the global response stage and the damping ratio is assumed to be 5%.

During an impact event, the beam is mobilized with the increased effective beam span l_{eff} 266 267 as stress waves propagate from the impact location to the boundaries as shown in Fig. 10. P-268 wave, shear wave and Rayleigh surface wave are generated in the beam. P-wave arrives the 269 supports first and then shear wave and Rayleigh wave comes last. P-wave causes the 270 longitudinal vibration while shear wave propagates transversely and causes the vertical 271 vibration of beam, which induces vertical reaction force and response. After reaching the 272 boundary, shear stress wave reflects and travels back to the impact location, then it is assumed 273 that the whole beam span is mobilized and the global response is activated in this study.

Therefore, the duration of local response stage t_0 can be determined by using the beam span land the velocity of shear stress wave v_s [7, 43] as follows:

$$t_0 = \frac{l}{v_{\rm s}} \tag{8}$$

276 The shear stress wave velocity v_s can be calculated by:

$$v_{\rm s} = \sqrt{\frac{E}{2\rho(1+\mu)}} \tag{9}$$

where *E*, ρ , and μ are Young's modulus, density, and Poisson's ratio of concrete. During the local response stage, it is assumed that the effective beam mass increases linearly as the increase of the effective beam span l_{eff} by considering the stress wave propagation. When the whole beam is mobilized in the global response stage after the duration of t_0 , the effective beam mass keeps constant [44] as follows,

$$m_{\rm b} = 0.493\rho bdl \tag{10}$$

where ρ is the density of an RC beam, *b* and *d* are the width and depth of beam section, respectively. In addition, the global stiffness of a beam k_b describing the relationship of resistance (F_b) and deflection (δ_b) is determined by beam section analysis as illustrated in section 4.1.2.



286

287

Fig. 10. Stress wave propagation and effective span of RC beam.

288 4.1.2. Beam section analysis

A nonlinear fiber beam section analysis of a pin-supported RC beam is conducted to determine the global stiffness of beam subjected to a concentrated load in this section. The relationship of moment (*M*) and curvature (ϕ) of beam section is calculated firstly by the nonlinear analysis. Then the relationship of resistance (*F*_b) and deflection (δ_b) is obtained by using the moment-curvature relationship (*M*- ϕ) and the resistance-deflection relationship (*F*_b- δ_b) is used to give the global behaviour of an RC beam under impact loads.





302 stress is expressed as a function of concrete strain as follows:

$$\sigma = \begin{cases} Kf_{c}' \left[2\left(\frac{\varepsilon}{\varepsilon_{0}}\right) - \left(\frac{\varepsilon}{\varepsilon_{0}}\right)^{2} \right] & (\varepsilon \leq \varepsilon_{0}) \\ Kf_{c}' \left[1 - Z(\varepsilon - \varepsilon_{0}) \right] & (\varepsilon_{0} < \varepsilon \leq \varepsilon_{u}) \end{cases}$$
(11a)

$$0.2Kf_{\rm c}'$$
 ($\varepsilon > \varepsilon_{\rm u}$)

$$\varepsilon_0 = 0.002K \tag{11b}$$

$$K = 1 + \frac{\rho_{\rm s} f_{\rm yh}}{f_{\rm c}'} \tag{11c}$$

$$Z = \frac{0.5}{\frac{3 + 0.29f_{\rm c}'}{145f_{\rm c}' - 1000} + 0.75\rho_{\rm s}\sqrt{\frac{h'}{s_{\rm h}}} - 0.002K}$$
(11d)

$$\varepsilon_{\rm u} = 0.004 + 0.9\rho_{\rm s}\left(\frac{f_{\rm yh}}{300}\right) \tag{11e}$$

where f_c is the concrete compressive strength, K is a variable considering the increase of compressive strength by stirrup confinement, ε_0 and ε_0 are the concrete strain corresponding to the maximum stress and the ultimate stress, ρ_s is the stirrup ratio and f_{yh} is the yield strength of stirrups, Z is the strain-softening slope, h' is the width of concrete core confined by stirrups, and s_h is the center spacing between stirrups. Moreover, a bilinear model is adopted to illustrate the tension behaviour of concrete as shown in Fig. 12(a). σ_{t0} and ε_{t0} are the tensile strength and strain of concrete, respectively.

310 The steel material model proposed by Esmaeily and Xiao is employed for steel rebar as311 shown in Fig. 12(b) [49]. The relationship of stress and strain for steel is given as follows,

$$\sigma = \begin{cases} E_{s}\varepsilon & (\varepsilon \leq \varepsilon_{y}) \\ f_{y} & (\varepsilon_{y} < \varepsilon \leq k_{1}\varepsilon_{y}) \\ k_{4}f_{y} + \frac{E_{s}(1-k_{4})}{\varepsilon_{y}(k_{2}-k_{1})^{2}} (\varepsilon - k_{2}\varepsilon_{y})^{2} & (\varepsilon > k_{1}\varepsilon_{y}) \end{cases}$$
(12)

312 where E_s , f_y , and ε_y are Young's modulus, yield strength, and yield strain of steel material,

313 respectively, k_1 , k_2 , k_3 , and k_4 are the parameters that control the shape of strain-stress curve as



314 presented in Fig. 12(b).

Fig. 12. Material models for concrete and steel.

To account for the strain rate effect on the material strengths, the DIFs for concrete and steel are employed [29, 30]. The strain rate of concrete and steel rebar can be calculated by using the curvature rate of beam section $\dot{\phi}$ as presented in Fig. 11.

Since the impact force can be deemed as a concentrated load acting on the RC beam at midspan, the simplified model for calculating the deflection δ of beam under concentrated load at midspan is illustrated in Fig. 13. The relationship between curvature ϕ of the beam section at midspan and deflection δ at midspan in the elastic and plastic stage [20, 21, 50] can be expressed as follows:

Elastic stage
$$\phi = \frac{12}{l^2} \delta$$
 for $0 < \delta \le \delta_y$ (13)

Plastic stage
$$\phi = \phi_{y} + \frac{4}{l \cdot l_{p}} (\delta - \delta_{y}) \text{ for } \delta > \delta_{y}$$
 (14)

where δ_y is the yield deflection, *l* is the beam clear span, and l_p is the length of the plastic hinge at midspan and can be calculated by the following equation proposed by Mattock [51],

$$l_{\rm p} = d + 0.05l \tag{15}$$

- 325 where *d* is the depth of beam section. Therefore, the relationships of curvature rate $\dot{\phi}$ of beam
- 326 section and deflection rate $\dot{\delta}$ (i.e., loading rate) are given as follows,

Elastic stage
$$\dot{\phi} = \frac{12}{l^2} \dot{\delta} \quad \text{for } 0 < \delta \le \delta_y$$
 (16)

Plastic stage
$$\dot{\phi} = \frac{4}{l \cdot l_{\rm p}} \dot{\delta} \quad \text{for } \delta > \delta_{\rm y}$$
 (17)

327





With the plane section assumption, i.e., the beam section remains plane under loading, the strain and strain rate distributes linearly along the beam depth as shown in Fig. 11. According to the curvature ϕ and curvature rate $\dot{\phi}$ of beam section, the strain ε_i and strain rate $\dot{\varepsilon}_i$ of concrete fiber or steel fiber can be derived as follows,

$$\varepsilon_i = \varepsilon_{\rm N} + \phi Z_i \tag{18}$$

$$\dot{\varepsilon}_i = \left(\frac{\varepsilon_{\rm N}}{\phi} + Z_i\right)\dot{\phi} \tag{19}$$

where ε_N is the section strain at the centroidal y-axis, Z_i is the coordinate of concrete fiber $(Z_{c,i})$ or steel fiber $(Z_{s,i})$ along z-axis of the beam section. The stress of concrete fiber $(\sigma_{c,i})$ and steel fiber $(\sigma_{s,j})$ can be calculated with the strain and the corresponding strain rate from the material constitutive model and DIF model. Based on the areas of section fiber (concrete fiber $(A_{c,i})$ and steel fiber $(A_{s,i})$) and stress of section fibers, the axial force (N) and bending moment (M) of the beam section are calculated by the equilibrium equations as follows,

$$N = \int \sigma dA = \sum_{i=1}^{m} \sigma_{c,i} A_{c,i} + \sum_{j=1}^{n} \sigma_{s,j} A_{s,j} = 0$$
(20)

$$M = \int \sigma Z_i dA = \sum_{i=1}^m \sigma_{c,i} Z_{c,i} A_{c,i} + \sum_{j=1}^n \sigma_{s,j} Z_{s,j} A_{s,j}$$
(21)

338 It should be noted that the axial force acting on the beam section is zero in this study. In order 339 to meet the equilibrium for axial force (N = 0) as expressed in Eq. (20), the section strain at the 340 neutral y-axis ε_N is determined by iterations. Once ε_N is obtained, the bending moment M can 341 be derived by Eq. (21) and the resistance F_b is calculated sequentially. To get the full section 342 response, the curvature ϕ increases at a step of 0.0004 1/mm as an iterative process. Fig. 14 343 illustrates the flowchart of beam section analysis to calculate the moment-curvature relationship $(M-\phi)$ and resistance-deflection relationship $(F_b-\delta_b)$ for RC beam. When the resistance-344 345 deflection relationship of RC beam is obtained by conducting the section analysis, the global stiffness k_b of RC beam shown in Fig. 9 can be determined accordingly. After defining all 346 347 parameters employed in the analytical model, the governing equations of 2DOF model as given 348 in Eq. (3) and Eq. (4) are solved by using the SciPy library in Python.





Fig. 14. Flowchart of fiber beam section analysis.

351 **4.1.3.** Calibration of the proposed analytical model

352 Two sets of RC beam tests are employed to calibrate the analytical model, i.e., RC beams 353 subjected to a concentrated load at midspan with various loading rates to calibrate the beam 354 section model, and RC beams under drop weight impact to check the validity of the developed 355 2DOF model. In terms of the first part of calibration (i.e. beam section analysis model), the RC 356 beams (S1616, S2222, and S1322) tested under different loading rates by Fujikake et al. [26] 357 are used. The dimension, rebar layout, and material strength of the RC beams are presented in 358 Fig. 2 and Table 1. A concentrated load with a loading rate of 0.5 mm/s (static loading) or 2000 359 mm/s (rapid loading) is applied at midspan of the RC beam and the relationship of the applied 360 load on beam and midspan deflection is captured. By using the beam section analysis as 361 illustrated in Section 4.1.2, the analytical load-deflection relationship is calculated and agrees 362 well with the test result for the specimens under different loading rates as shown in Fig. 15. In 363 addition, it is noted that there are some fluctuations in the test results under a higher loading 364 rate of 2000 mm/s as presented in Fig. 15(b), which is not observed in the analytical results. 365 This is because there are local vibrations between the loading actuator and RC beam at the 366 loading point in the test, while the local interaction is not considered in the beam section 367 analysis. In general, the beam section analysis results are in good agreement with the test results, 368 indicating that the beam section analysis model can be used to calculate the global stiffness of beams in the proposed 2DOF model. 369



analysis.

In addition, the 2DOF analytical model is calibrated by the test data of RC beams under drop weight impact tests [22, 26, 52]. The detailed information of the geometry of beam, rebar layout, and impact energy of the impact tests were reported in the literature. It is worth noting that the contact areas in these tests are different due to different geometries of drop weight head and beam section, causing various contact stiffness $k_{\rm b}$. The drop weight head is hemisphere with a radius of 90 mm in Ref. [26] while a flat contact surface with a diameter of 200 mm is employed in the drop weight tests on the beams with width of 150 mm in Ref. [52] and 200 mm

in Ref. [22], respectively. Therefore, the smaller contact area in Ref. [26] results in a lower contact stiffness than that in Refs. [22, 52]. The contact stiffness for the tests in the Refs. [22, 26, 52] are determined as 0.8×10^8 N/m, 10×10^8 N/m, and 15×10^8 N/m, respectively, by using Eq. (5) and Eq. (6). The predicted impact force and displacement at midspan are in good agreement with the test results as shown in Fig. 16 (a) and (b). The impact force profile is well predicted as presented in Fig. 16(a) by the analytical method, indicating that the developed 2DOF model can be used to calculate the impact force profile of pin-supported RC beams under

384





Fig. 16. Comparisons of test results [22, 26, 52], analytical predictions, and results predicted by equations (22)~(27) of RC beams under drop weight impact.

4.2. Development of the simplified impact force profile model

386	Based on the impact force profiles recorded in test [4, 22, 26, 52, 53], a simplified impact
387	force profile is suggested as shown in Fig 17. It is defined by six characteristic points (OABCDE)
388	and composed of a primary triangular area (OAB) with the peak impact force F_a and trapezoidal
389	force block (BCDE) with a constant plateau force F_p . By considering the key parameters
390	affecting the impact force profile as summarized in Fig. 1, the force (F_a and F_p) and duration
391	$(t_a, t_b, t_{cd}, and t_e)$ of the characteristic points as presented in Fig 17 can be determined by using



396 A total of six key parameters affecting the impact force profile are employed in this 397 analytical study as listed in Table 3. The values marked in red for each parameter are the reference value for the analytical parametric study, that is, in each case only one parameter is 398 399 changed while other parameters adopt the reference values. Without loss of generality in this 400 study, the values for the six parameters used to fit empirical equations are chosen from the widely used design parameters in drop weight impact tests as given in Table 3. A total of 59 401 402 specimens considering different combinations of parameters listed in Table 3 are investigated 403 to obtain the impact force profile by using the calibrated 2DOF analytical method. Once the

impact force profile of each specimen is generated from the analytical results, the characteristic points illustrated in Fig. 17 are determined and collated into a database. Through multivariable regression analysis, six empirical equations $(22)\sim(27)$ are proposed to predict the impact force and duration of characteristic points by using the parameters with their own dimensions. It should be noted that the beam section flexural stiffness *EI* (10^6 N·m²) is included in these equations to quantify the effect of the beam section on the global stiffness and plateau force. If the empirical equations (22)~(27) are used to predict the impact force profile, the six parameters

411 should be chosen within the applicable range with the specified units as listed in Table 3

$$F_a = 302.02k_c^{0.588} + 0.78m_d^{0.5} + 62.88v_0 - 259.49 \qquad (R^2 = 0.94) \qquad (22)$$

$$F_{\rm p} = 19.97k_{\rm c}^{0.26} + 15.82\nu_0^{0.32} + 160.22l^{-0.9} + 47.25\ln(EI) + 41.18\rho - 185 \qquad (R^2 = 0.93)$$
(23)

$$t_a = 0.84 + \frac{0.112}{k_c} + \frac{0.148}{v_0}$$
 (*R*² = 0.82) (24)

$$t_b = e^{(-0.08k_c + 0.0002m_d + 0.0046v_0 + 1.2)}$$

$$(R^2 = 0.75)$$

$$(25)$$

$$(0.00123m_s + 0.135v_s + 0.348l - 0.047El - 0.80 + 2.55)$$

$$(R^2 = 0.75)$$

$$(26)$$

$$t_{cd} = e^{(0.00125m_d + 0.135\nu_0 + 0.348i - 0.047E1 - 0.89 + 2.55)} \qquad (R^2 = 0.88) \qquad (26)$$

$$t_e = e^{(0.001m_d + 0.100 + 0.307t - 0.030Lt - 0.333p + 2.993)}$$
 (*R²* = 0.89) (27)

412 **4.3. Verification of the simplified impact force profile model**

To verify the proposed equations $(22)\sim(27)$ for predicting the impact force and duration of the simplified impact force profile, a total of 39 sets of tested data including experimental results [22, 26, 52, 53] and numerical results in section 3.2 are used for validation. All the 39 sets of data have the Type III impact force profile and the failure models are global flexure or flexure-shear failure instead of severe local shear failure. Fig. 18 shows the comparison of the test data and predicted results of the characteristic points. The coefficient of correlation *R* between test data and predicted results and the mean value of predicted-to-tested ratio (Mean)

are given. Fig. 18(a)~(b) show that the predicted impact forces (F_a and F_p) present good correlations with the test data by yielding R = 0.984 and Mean = 0.998 for the peak force, and R = 0.889 and Mean = 1.028 for the plateau force. Besides, the predicted durations as shown in Fig. 18(c)~(f) also agree well with the test data. It is noted that the coefficient of correlation Rof the peak force time t_a and primary duration t_b are relatively lower than those of force plateau duration t_{cd} and total impact duration of t_e because the linear behaviour of the contact stiffness

426

is employed in this analytical study.





427 Based on the impulse-momentum conservation theorem, the impulse is equal to the change 428 of the momentum during impact. Upon drop weight impact, the beam at midspan reaches the 429 maximum displacement and the velocities of beam and drop weight are zero at point D as shown 430 in Fig 17. Therefore, the impulse calculated by the area of *OABCDF*, i.e., the grey area in Fig 431 17 should be equal to the initial momentum of drop weight. In order to further verify the 432 proposed equations for impact force profile, the initial momentum of drop weight and predicted 433 impulse are compared as shown in Fig. 19. The predicted impulse applied on the beam is 434 comparable to the initial momentum of drop weight, with the correlation value R = 0.878 and 435 Mean = 1.252.



436 437

Fig. 19. Comparison of initial momentum and predicted impulse.

In addition, the simplified impact force profiles predicted by the proposed equations are also compared with experimental results as shown in Fig. 16(a). It is observed that the predicted simplified impact force profiles are in good agreement with the tested impact force profile in general. It is concluded that the proposed equations can well predict the simplified impact force profile for pin-supported RC beams under drop weight impact.

443

4.4. Discussion on the applicability of the proposed analytical model

444 The 2DOF analytical model for pin-supported RC beams under drop weight impact is 445 developed and the empirical formulae for predicting the simplified impact force profile are also 446 proposed based on test and analytical results. The proposed formulae give a good prediction of the overall impact force profile and can rapidly predict the simplified impact force profile for 447 448 the design of drop weight impact tests. However, since only the global stiffness of RC beams is 449 considered in the analytical derivation while the local shear stiffness of RC beams is ignored, 450 the proposed simplified impact force profile model is only applicable for the RC beams with 451 global flexure or flexure-shear deformation under impact. Therefore, the failure mode of RC 452 beams under impact should be examined first to determine the applicability of the proposed 453 models. Whether the beam experiencing shear failure depends on the characteristic of impact 454 loading (duration and amplitude) and the impact load-carrying capacity of beam [54-57]. If the 455 impact loading amplitude is high and the ratio of $T_{\rm m}/T_{\rm M}$ is very small, the failure mode of beam 456 is governed by shear or/and punching shear failure [54, 55]. In addition, the beam subjected to 457 an impact loading with peak force higher than the dynamic shear load-capacity of beam section 458 is prone to experience punching shear failure [56, 57]. Diagonal shear cracks occur along the failure section with an approximate angle of 45° [5, 8, 56] as shown in Fig. 20. Therefore, the 459 460 failure mode of beam can be pre-determined by comparing the peak impact force and the dynamic shear capacity P_{dyn}^{max} of the beam section. 461





462 463

Fig. 20. Local punching shear failure of beam section.

464 The dynamic shear capacity P_{dyn}^{max} of RC beams can be estimated by using the simplified 465 equation proposed by Do et al. [57] as follows:

$$P_{\rm dyn}^{\rm max} = 6.5 \times \frac{f_c'}{10} \times b \times d \tag{28}$$

where *b* and *d* are the width and depth of beam section, f_c is the concrete compressive strength. Local punching shear failure appears when the peak impact force reaches P_{dyn}^{max} . Therefore, to determine the failure mode of RC beam, the peak impact force can be firstly estimated by Eq. (22) and then compared with P_{dyn}^{max} . If the calculated peak impact force is less than the dynamic shear capacity P_{dyn}^{max} of beam section, it is deemed that RC beam presents global deformations instead of severe local punching shear failure and the proposed equations (22)~(27) can be used to predict the impact force profile of RC beam.

473 **5.** Conclusion

The impact force profile of RC beams under drop weight impact is numerically and analytically investigated in this study. The factors affecting the impact force profile are summarized and the influence of the global stiffness on the impact force plateau is quantified. A 2DOF analytical model is developed to predict the impact force profile with good accuracy. The simplified impact force profile can be straightforwardly estimated by using empirical equations (22)~(27). The main conclusions can be drawn as follows.

The global stiffness governs the impact force plateau but has a negligible effect on the peak
impact force. Decreasing the global stiffness of RC beam would weaken the interaction
between the drop weight and the beam after the primary impulse, causing a decrease of

483 plateau force and an increase of plateau stage duration.

484	2.	A 2DOF analytical model is developed for the pin-supported RC beams under drop weight
485		impact, and gives accurate predictions of impact force and beam displacement response.
486	3.	Intensive analytical calculations are carried out to predict the impact force profiles, and the
487		results are used together with available testing data to develop empirical formulae for
488		predictions of impact forces. The proposed empirical formulae are proven yielding good
489		predictions of impact force profiles that would be induced by drop weight on RC beams.
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